

The background of the slide is a high-resolution image of the Earth as seen from space. The curve of the planet is visible, showing a mix of blue oceans, white clouds, and dark landmasses. The lighting creates a bright horizon line, giving the Earth a glowing appearance against the blackness of space.

**Current and future Earth
Observation missions:
an Italian data portfolio unique in
the world .**

ASI Earth Observation at glance: *for our planet, for our future...*

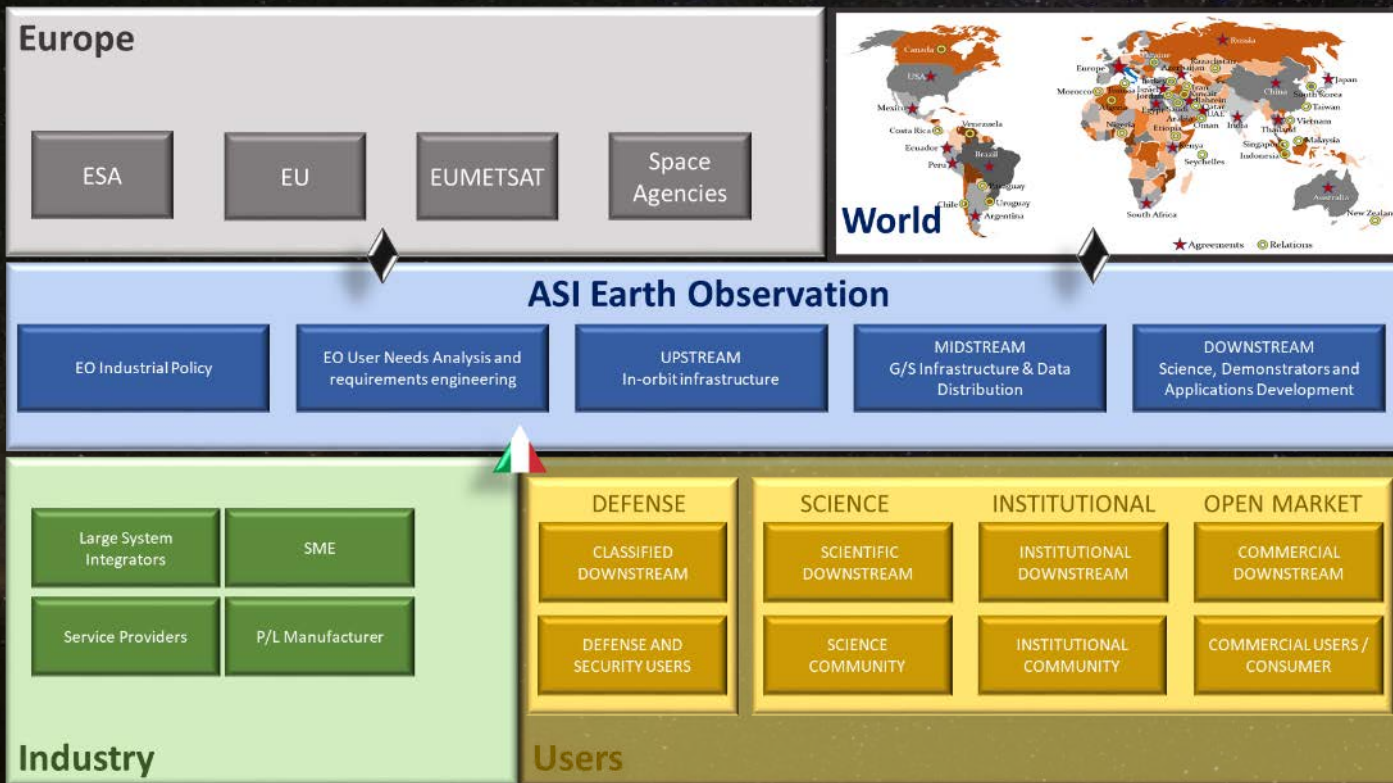


Agenzia Spaziale Italiana

...ensure the understanding, monitoring and protection of our planet guaranteeing the maximum benefit for science, applications, society and economic growth of Italy...

ASI implements this vision through its Earth observation programs developed in Italy, in Europe (ESA, EU, EUMETSAT) and within the widest international framework.

ASI plays an architect role in the definition, development, procurement, exploitation and evolution of the space assets through its National industry and science and application community, contributing actively to the major-European infrastructures.



Our 8 major objectives in Earth Observation

Sustain the Future of Synthetic Aperture Radar

- New instruments and constellations (X/L/P Bands)
- Developments for large reflectors/deployable antennas
- SAR Technology Roadmap

Secure the leadership in Hyperspectral payload

- Hyperspectral Next Generation
- Miniaturized Hyperspectral Mission (PLATINO)
- Hyperspectral Technology Roadmap

Strengthen developments in Thermal Infrared

- TIR Mission based on minisatellite (PLT-2)
- ASI-NASA TIR mission
- TIR hyperspectral / TIR Technology Roadmap;

Achieve autonomy in HR systems

- HR Mission based on minisatellite (Feasibility study)
- Optical Technology Roadmap

Consolidating the Lidar capability

- Lidar mission (Feasibility study)
- Lidar Technology Roadmap

Sustain development of new instruments

- Radiometers, Quantum Gravimetry, etc
- Technology roadmap

Strengthen Earth science and applications

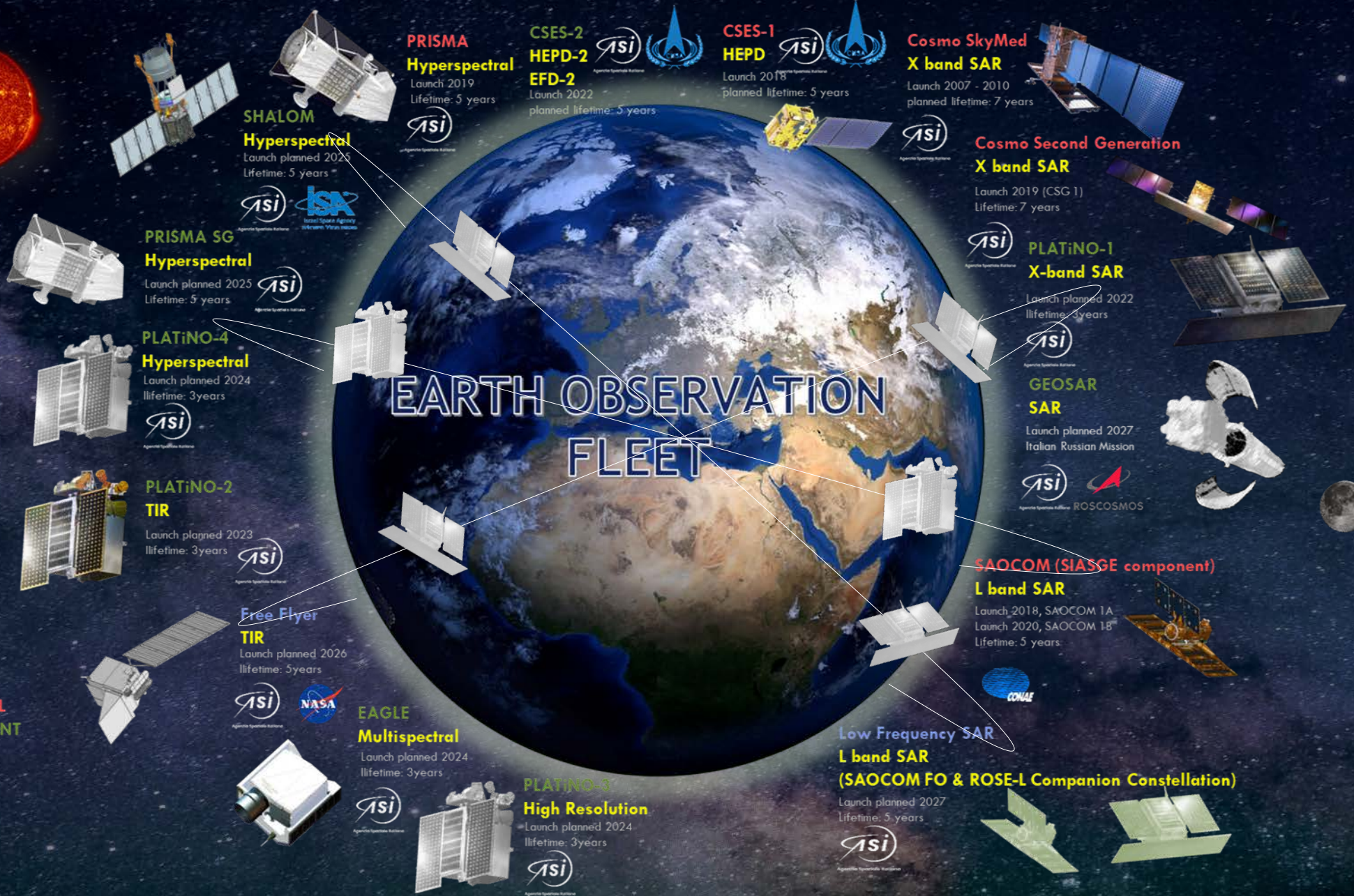
- Scientific and application projects 1) to understand earth system and interaction between process and 2) to deliver information to enhance quality of life and strengthen our economy

Pull users towards our applications and services layers

- Demonstrators
- Application Services Start-Up

EARTH OBSERVATION FLEET

OPERATIONAL
IN DEVELOPMENT
PLANNED



THE REFLECTIVE/EMISSIVE BANDS: VIS-NIR-SWIR-TIR

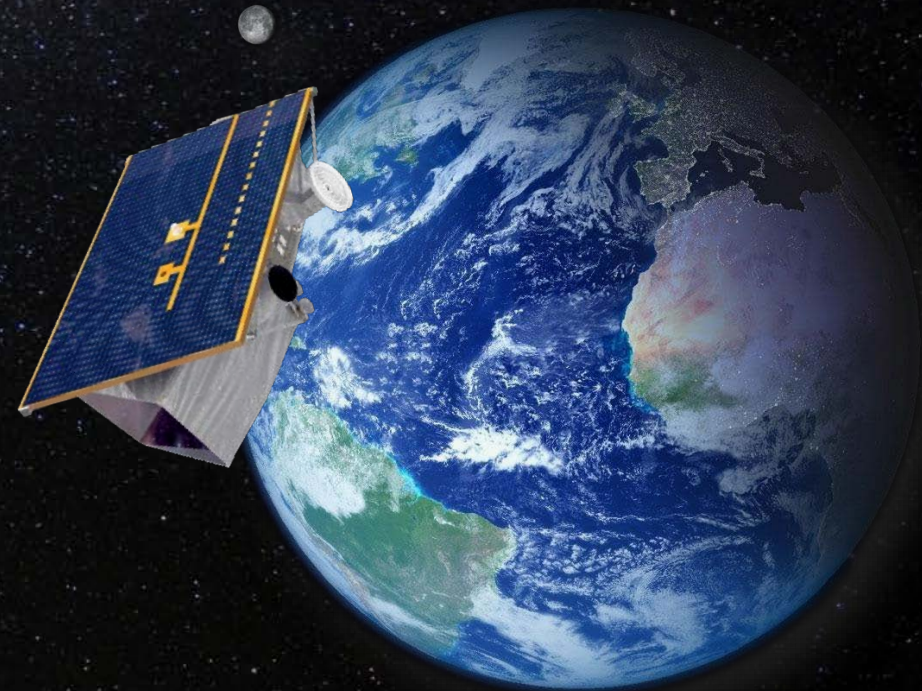
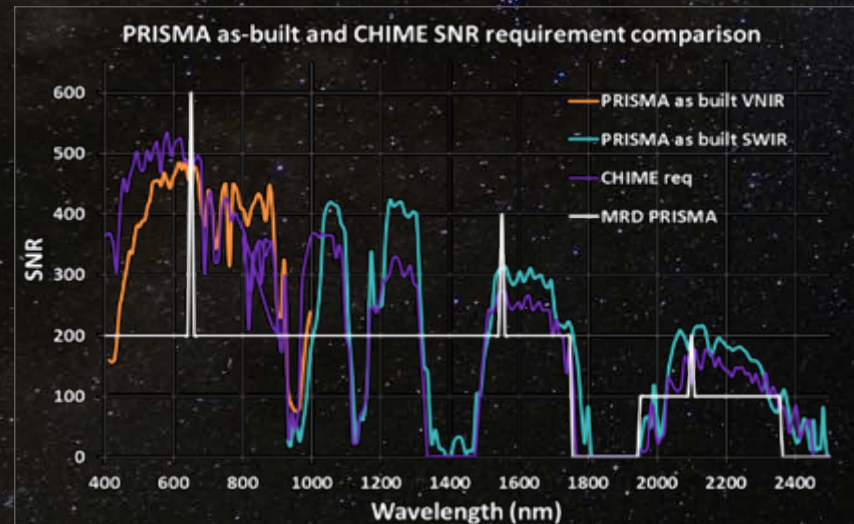
Focus on:

- Hyperspectral imagery in the visible and shortwave infrared;
- Multi / hyperspectral imagery in the thermal IR.

PRISMA - Hyperspectral

Fully funded by the Italian Space Agency (ASI): in-orbit Earth Observation system that simultaneously combines data of a hyperspectral sensor and a panchromatic camera from the same scene.

- 240 total bands in VNIR (#66, 400–1010 nm) & SWIR (#174, 920–2505 nm) at a spatial resolution of 30 m on a swath of 30 km
- Mean spectral resolution of 10 nm in a spectral range of 400-2500 nm
- Pan (Panchromatic) imagery is provided at a spatial resolution of 5 m
- Simultaneous acquisition of images in the VNIR, SWIR and PAN on the same scene!



PRISMA 2nd Generation

PRISMA Second Generation is the future Hyperspectral Italian Mission, to be launched in 2025.

- Entirely Funded by the Italian Space Agency
- Hyperspectral data continuity currently available by the PRISMA system.

SPECS:

- SWATH and SNR: on-demand techniques of SWATH enlargement and SNR enhancement on a single pass using the platform agility.
- Acquisition modes: STRIPMAP and SPOTLIGHT.
 - STRIPMAP image: VNIR/SWIR GSD ≤ 30 m and PAN GSD ≤ 5 m, swath ≥ 30 km and indefinite length with a Daily STRIPMAP Imaging Capacity (acquire, downlink and archive) more than 2.000.000 km².
 - SPOTLIGHT image (on-demand): VNIR/SWIR GSD ≤ 10 m and PAN GSD $\leq 2,5$ m, swath ≥ 30 km and length up to 210 km with a Daily SPOTLIGHT Imaging Capacity (acquire, downlink and archive) more than 200.000 km².
- Low revisit time (72 h with a maximum off-nadir angle of $\pm 30^\circ$)



SHALOM: Spaceborne Hyperspectral Applicative Land And Ocean Mission

Joint program between ASI and ISA based on the "Implementation Arrangement On Cooperation in a Joint Definition Phase of a Spaceborne Hyperspectral Applicative Land And Ocean Mission - Shalom".

Italy is responsible of the overall hyperspectral instrument and is the Ground Segment Authority.

Israel is responsible for the satellite platform, the telescope and the panchromatic camera and is the Space Segment Authority.

The mission will acquire:

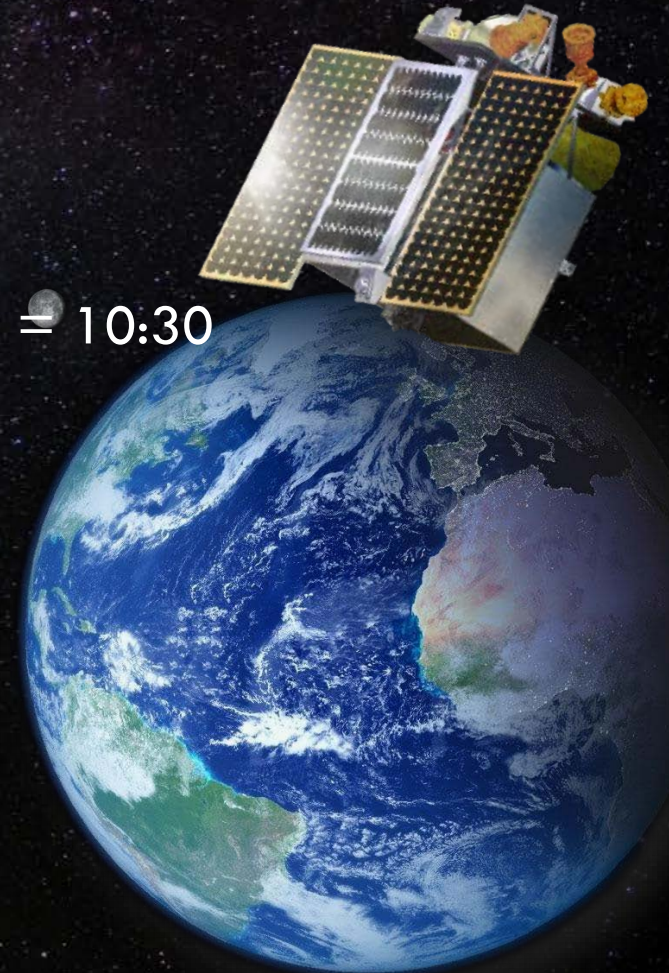
- ⦿ Hyperspectral measurements of the Earth radiation reflection and absorption in the VIS-NIR-SWIR spectral range (400-2500nm)
- ⦿ Panchromatic images of the Earth (simultaneous with the hyperspectral measurements)

| Parameter | Specification |
|---------------|-------------------------------|
| Orbit | SSO 640 km |
| Repeat Cycle | 4 days |
| Daily Imaging | Up to 200.000 km ² |
| Swath Width | ≥ 10 km |
| HYP P/L GSD | ≤ 10m |
| PAN P/L GSD | ≤ 5m |
| Op Lifetime | 5 years |

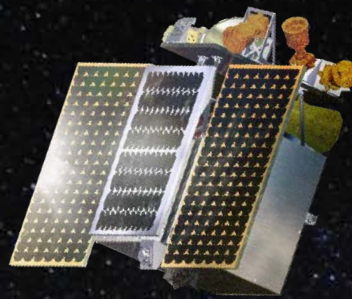


PLATiNO – 2: TIR/Multispectral Mission

- Spectral Range 8-12 micron
- Spectral Channels : 8.6, 9.1, 10.3, 11.5 micron
- Channel bandwidth: 1 micron
- Spatial resolution: 40m
- Accuracy $< 1.5^{\circ}$ K
- Swath = 40 km
- Strip = up to 170 km
- Daily coverage 170.000 km²
- Secondary P/L:
 - VNIR camera
 - Early Warning system
- Orbital parameters:
 - SSO Frozen
 - Local time of ascending node = 10:30
 - Altitude = 393 km
 - Inclination = 96 deg
 - Repeat Cycle = 52 days



PLATiNO – 3



High Resolution Mission

Planned Launch date 2024

| | |
|------------------------|--------------------------|
| band | VIS |
| Altitude [km] | 400 |
| Orbital duty cycle [s] | 180 to 300 |
| Swath [km] | 4 |
| FOV [deg] | 1.27 |
| Aperture [mm] | 420 |
| GSD [m] | 0.5 x 0.5 |
| Spectral Range (nm) | 470-840 |
| VIS SNR | 100-120:1 |
| MTF | 0.2-0.3 (over RGB bands) |
| # spectral bands | 4 (RGB-NIR) |

PLATiNO-4



Hyperspectral Mission

Planned Launch date 2025

| | |
|--------------------------|---|
| band | VNIR - SWIR |
| Altitude [km] | 619 |
| Orbital duty cycle [s] | 65 |
| Swath [km] | 30 |
| FOV [deg] | 2.77 |
| Aperture [mm] | 210 |
| GSD [m] | 30 |
| Spectral Range (nm) | VNIR: 400 – 1010 / SWIR: 920 – 2505 |
| Spectral Resolution (nm) | 10 |
| VNIR SNR | > 200:1 |
| SWIR SNR | > 100:1 |
| MTF | VNIR/SWIR along track > 0.18 / VNIR/SWIR across track > 0.34 / PAN along track > 0.10 / PAN across track > 0.20 |
| # spectral bands | >230(VNIR-SWIR) |

ASI-NASA/JPL: TIR-Multispectral Mission

ASI-JPL FreeFlyer mission is a TIR-VNIR Mission aimed at acquiring images of the Earth 24/7



TIR instrument: 8-band radiometer

| TIR Performance | SPECS | Note |
|-------------------|--|--------------------------------|
| Thermal IR Bands | 8.28 μ m / 8.63 μ m / 9.07 μ m / 11.33 μ m / 12.05 μ m | |
| mid-IR bands | 3.98 μ m / 4.80 μ m | |
| short-wave IR | 1.60 μ m | |
| NETD | 0.2 °K @ 300 °K | |
| GSD | 60m | |
| Swath width | 935 km | |
| Coverage | Global | |
| Data quantization | 16 bit | |
| Data Production | 248.8 Mbps peak (108.17 Mbps compressed) | Daytime and Coastal Land (42%) |
| Data compression | > 2:1 | 2.3:1 used |
| Data Latency | < 24h | |

VNIR camera: a two-channel instrument to calculate NDVI values

| VNIR Performance | SPECS | Note |
|-------------------------|--|----------------------------------|
| Visible Bands center | 655 nm | |
| Visible Bands bandwidth | 80 nm | |
| NIR Bands center | 835 nm | |
| NIR Bands bandwidth | 80 nm | |
| SNR | 100 | For both bands |
| GSD | <35 m | |
| Swath width | 935 km | |
| Coverage | Global | |
| Data quantization | 12 bit | |
| Data Production | 133.8 Mbps peak (44.6 Mbps compressed) | Daytime only /Coastal Land (42%) |
| Data compression | 3:1 | 2.3:1 used |



THE MICROWAVES:

SAR in P, L, (C) and X Band

Focus on:

- COSMO-SkyMed: First, Second and Next Generations
- SAOCOM, L-Band
- GEOSAR, a GEOSYNCRONOUS SAR MISSION
- PLATiNO-1: MONO/BI STATIC X-BAND SAR MISSION
- P-Band, SAR and Sounder

COSMO-SkyMed: The First and the Second Generation

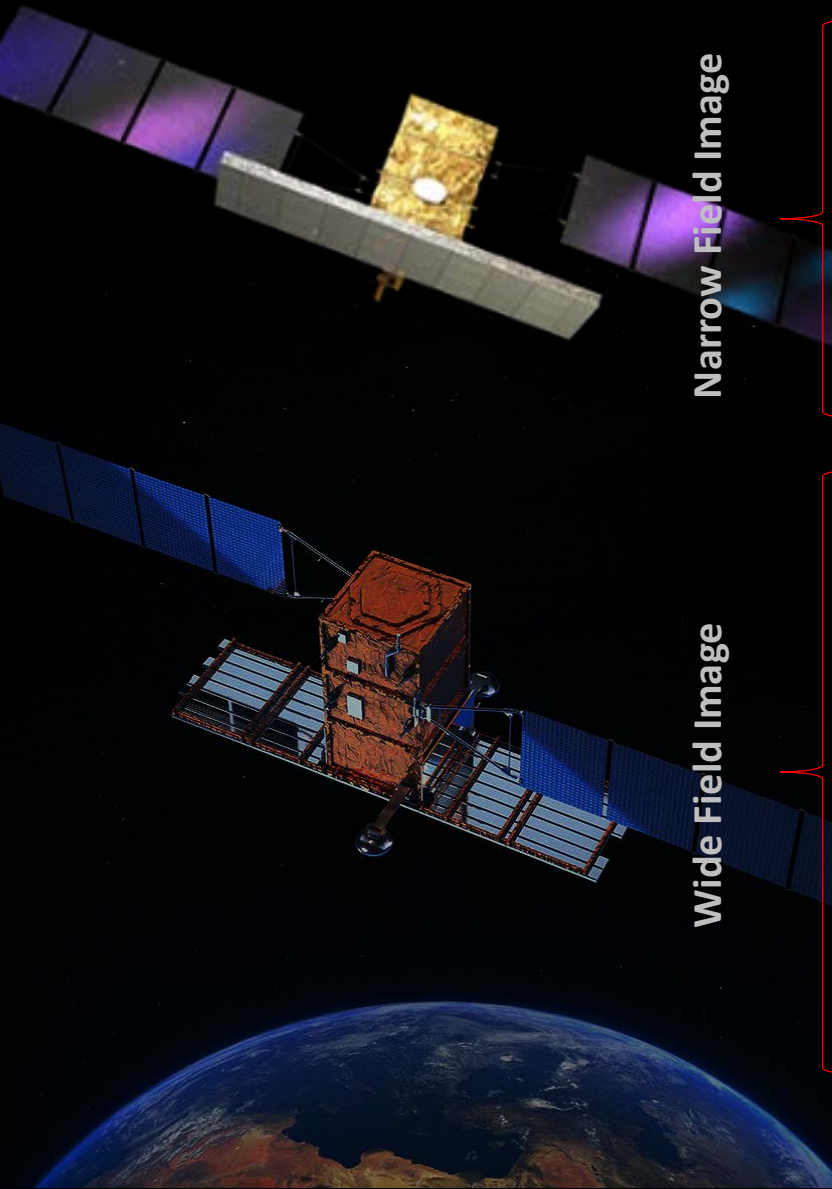
COSMO-SkyMed Second Generation (CSG) will:

- Ensure operational continuity to the currently operating constellation
- Achieve a step ahead in terms of functionality, performances and system services for the Earth Observation users

The 4 CSG Satellites will have an operational lifetime of at least 7 years.



CSG Image Products improvement w.r.t. CSK



Narrow Field Image

Wide Field Image

**Very High Resolution
VHR (sub-metric)**
Governmental Use

**Resolution: 1 m
Single Polarization
Size 10 km x 10 km**
Civilian and Defence use

**Resolution: 3 m
Single Polarization
Swath Size 40 km**
Civilian and Defence use

**Resolution: 30 m
Single Polarization
Swath Size: 100 km
or
Resolution : 100 m
Single Polarization
Swath Size: 200 Km**
Civilian and Defence use

Ultra-High Resolution (UHR)
Governmental Use

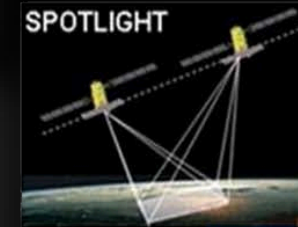
**Spot-2
VHR and Dual Pol. (**)**

Sp-2A res. $\leq 0.35 \times 0.55$ m
Swath $\geq 3.1 \times 7.3$ Km
Sp-2B res. $\leq 0.63 \times 0.63$ m
Swath $\geq 10 \times 10$ Km
Sp-2C res. $\leq 0.80 \times 0.80$ m
Swath $\geq 5 \times 10$ Km
Civilian and Defence Use

**Resolution : 3m x 3m
Swath Size Dual Pol 40 km
Swath Size QUADPOL 15 km**
Civilian and Defence use

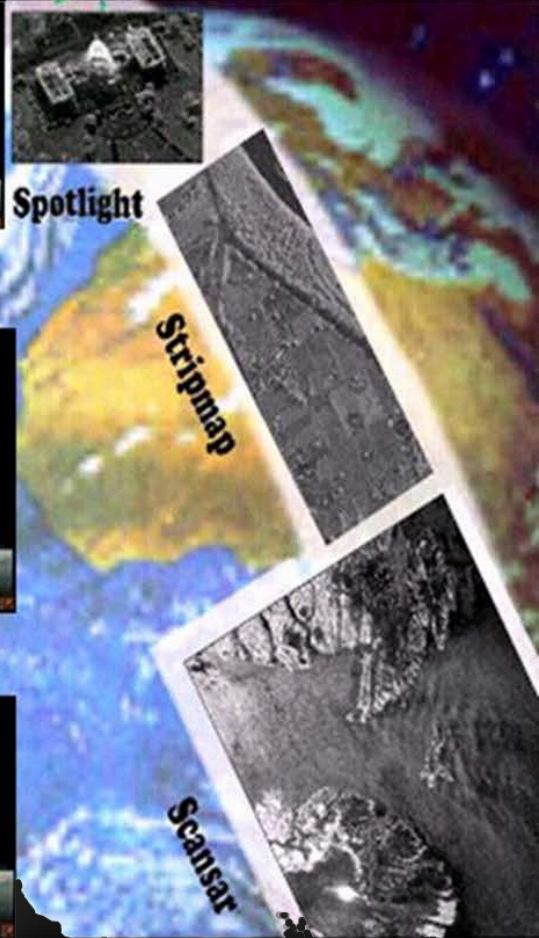
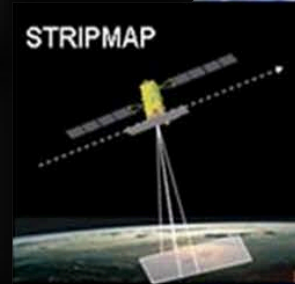
**Resolution : 4 x 20 m
Double Polarization
Swath Size : 100 km
or
Resolution : 6 x 40 m
Double Polarization
Swath Size: 200 Km**
Civilian and Defence use

SPOTLIGHT



Spotlight

STRIPMAP



Stripmap

SCANSAR



Scansar

(**) in azimuth and range

PLATiNO-1 – SAR Mission

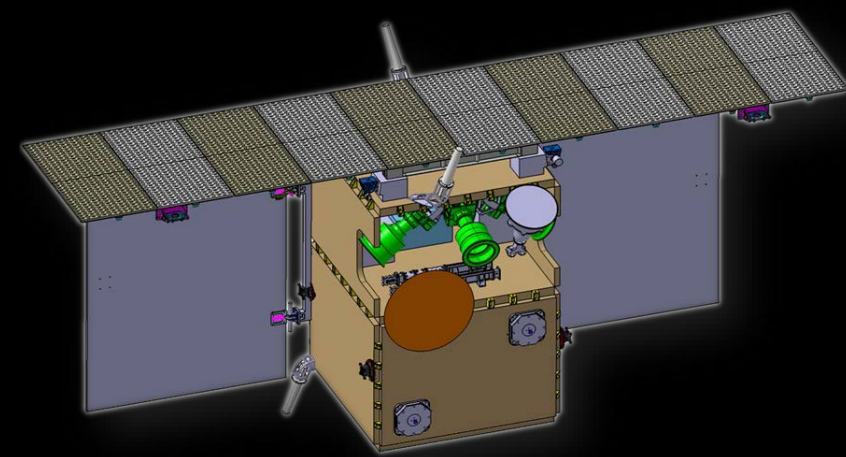
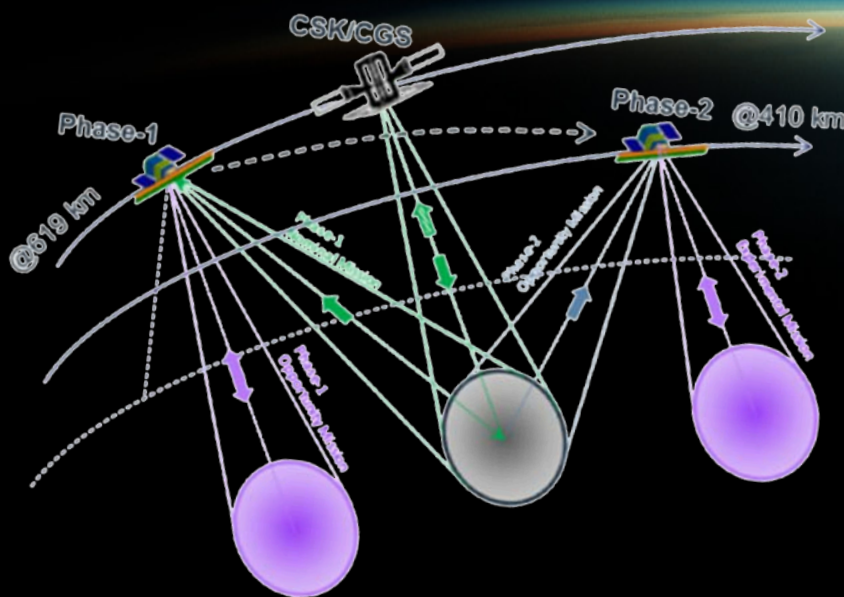
Mission Phases are:

- Commissioning (LEOP and Commissioning) 3 months;
- Phase-1 (@619 km, formation flying with CSK/CSG) 1 year;
- Re-orbit phase (orbit transfer with HET) 6 months;
- Phase-2 (@410 km, monostatic acquisition) 1.5 years;
- De-orbiting phase 6 months.

Selectable Formation-Flying configurations:

- Leader-Follower
- Pendulum
- Cartwheel
- Helixmonths

PLT-1 shall be sized to provide the capacity to acquire, downlink and archive images totaling **20000 km² daily**.



During Phase-1 PLT-1 will mainly work as a receiver acquiring from Earth the signal generated by CSK/CSG

| Bistatic performances (Phase-1) | |
|---------------------------------|------------------|
| Altitude | 619 km |
| Swath | 40 km |
| Resolution | 3 m |
| Target Experimental Resolution | 1 m |
| Imaging mode | CSK/CSG Stripmap |
| Continuous stripmap | Up to 1000 km |

| Monostatic performances (Phase -2) | |
|------------------------------------|-------------|
| Altitude | 410 km |
| Swath | 15 km |
| Resolution | 3 m |
| Target Experimental Resolution | 1 m |
| Imaging Mode | Stripmap |
| Continuous stripmap | Up to 800km |

GEOSAR – GEOSYNCHRONOUS SAR MISSION

Geosynchronous space-borne SAR system in cooperation with ROSCOSMOS

The mission will generate SAR products up to L1D level and perform interferometric co-registration of acquisitions with interferograms and coherence maps.

PRODUCTS:

- ⦿ Quicklook of the full image (for catalogue/data structure)
- ⦿ Tropospheric delay maps
- ⦿ Speckle filtered
- ⦿ Mosaicked images
- ⦿ Coregistered images
- ⦿ Interferograms / Coherence maps



SAOCOM – L-band polarimetric SAR

SAOCOM-SAR is an L-band polarimetric SAR instrument, the prime payload of the mission providing all weather, day/night observations to satisfy most of the applications considered in the Argentinean National Space Program, involving studies on agriculture, fishery, forestry, weather, hydrology, oceanography, emergencies, natural resources of land and sea, urban areas, cartography, geology, mining, soil exploitation, archeology and health

| Parameter | Value | Parameter | Value |
|------------------------------------|-----------------------|---|---|
| Center frequency | 1275 MHz (L-band) | Antenna looking angle | left or right side of path (left side is default) |
| | | Incidence angles | 20-50° |
| Maximum bandwidth | <45 MHz | Data quantization | 8 bit |
| Transmit peak power | 3.1 kW | Duty cycle | 15% (about 15 min/orbit) |
| Operational modes | Stripmap TopSAR | Stripmap high resolution TopSAR wide mode | 10 m x 10 m (pixel) 100 m x 100 m (pixel) |
| Stripmap swath width | > 65 km (each beam) | ScanSAR wide swath TopSAR narrow swath | > 320 km > 170 km (quadpol) |
| Signal transmission | HH or VV polarization | Signal reception Reception (double polarization) | HH or VV (single pol.) HH & HV or VV & VH |
| NESZ (Noise Equivalent Sigma Zero) | <-25 dB | Stripmap mid-resolution TopSAR narrow mode | 25 m x 25 m (pixel) 50 m x 50 m (pixel) |



P-Band – Sounder / SAR



Aerial radar multi-operating/multi-frequency modality in the UHF and VHF bands: the radar system operates at different carrier frequencies as Sounder and Synthetic Aperture Radar (SAR):

- Sounder operates at 165 MHz,
- SAR operates at
 - 450 MHz (SAR-Low mode)
 - 860 MHz (SAR-High mode).

Several Helicopter-Borne Campaign: 2 in Southern Italy and 1 in Morocco Desert

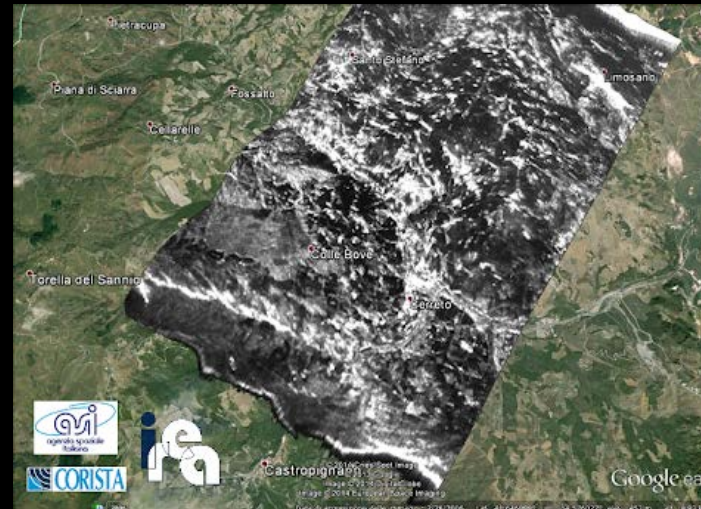
Low Frequency RADAR Mission

Objective:

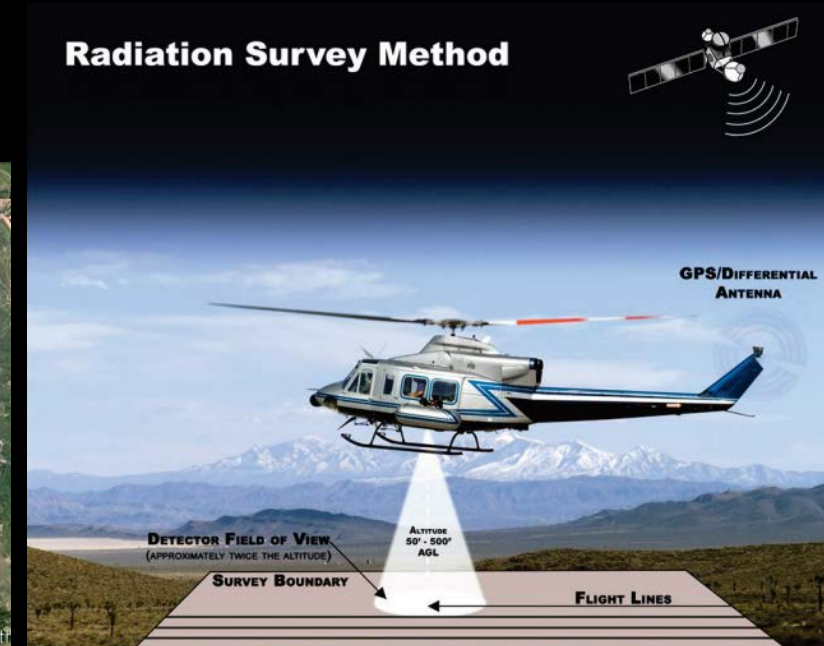
To Explore, at national level, the feasibility of a SAR mission in P (below 1 Ghz) and L bands.

The initiative capitalizes:

- the SAOCOM experience;
- the know-how matured in the P-Band experimental activities.



Radiation Survey Method



ASI EO future mission

- ◎ EO constellation with high temporal resolution;
- ◎ Mono and Bi-Static SAR X/L/P band, including COSMO-SkyMed Next Generations, Small-sat and Geosynchronous;
- ◎ Hyperspectral (VNIR/SWIR/TIR);
- ◎ LIDAR;
- ◎ Optical HR;
- ◎ New instruments: Radiometers, Quantum Gravimetry, Radar sounder etc..

*Supporting Earth science and applications and
pulling Users towards our services layers*





Agenzia Spaziale Italiana

THANK YOU FOR YOUR ATTENTION

